

AD-A095 344

TECHNICAL  
LIBRARY

AD

TECHNICAL REPORT ARLCB-TR-80047

FINITE ELEMENTS FOR INITIAL VALUE PROBLEMS IN DYNAMICS

T. E. Simkins

December 1980



US ARMY ARMAMENT RESEARCH AND DEVELOPMENT COMMAND  
LARGE CALIBER WEAPON SYSTEMS LABORATORY  
BENÉT WEAPONS LABORATORY  
WATERVLIET, N. Y. 12189

AMCMS No. 611102H600011

DA Project No. 1L161102AH60

PRON No. 1A0215601A1A

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED

#### DISCLAIMER

The findings in this report are not to be construed as an official Department of the Army position unless so designated by other authorized documents.

The use of trade name(s) and/or manufacturer(s) does not constitute an official indorsement or approval.

#### DISPOSITION

Destroy this report when it is no longer needed. Do not return it to the originator.

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER ARLCB-TR-80047	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) FINITE ELEMENTS FOR INITIAL VALUE PROBLEMS IN DYNAMICS		5. TYPE OF REPORT & PERIOD COVERED
		6. PERFORMING ORG. REPORT NUMBER
7. AUTHOR(s) T. E. Sinkins		8. CONTRACT OR GRANT NUMBER(s)
9. PERFORMING ORGANIZATION NAME AND ADDRESS Benet Weapons Laboratory Watervliet Arsenal, Watervliet, NY 12189 DRDAR-LC3-TL		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS AMCMS No. 611102H600011 DA Project No. 1L161102AH60 PRON No. 1A0215601A1A
11. CONTROLLING OFFICE NAME AND ADDRESS US Army Armament Research & Development Command Large Caliber Weapon Systems Laboratory Dover, NJ 07801		12. REPORT DATE December 1980
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		13. NUMBER OF PAGES 27
		15. SECURITY CLASS. (of this report) UNCLASSIFIED
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report)  Approved for public release; distribution unlimited.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES To be published in American Institute of Aeronautics & Astronautics (AIAA).		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Finite Elements Dynamics Boundary Value Problems Approximations		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) The work of C. D. Bailey amply demonstrates that a variational principle is not a necessary prerequisite for the formulation of variational approximations to initial value problems in dynamics. While Bailey successfully applies global power series approximations to Hamilton's Law of Varying Action, the work herein shows that a straightforward extension to finite element formulations fails to produce a convergent sequence of solutions. The source of the (CONT'D ON REVERSE)		

20. Abstract (Cont'd)

difficulties and their elimination are discussed in some detail and a workable formulation for initial value problems is obtained. The report concludes with a few elementary examples showing the utility of finite elements in the time domain.

## TABLE OF CONTENTS

	<u>Page</u>
ACKNOWLEDGMENTS	ii
INTRODUCTION	1
FINITE ELEMENTS IN TIME	2
HAMILTON'S PRINCIPLE - A CONSTRAINED VARIATIONAL PRINCIPLE	4
GLOBAL AND PIECEWISE RITZ APPROXIMATIONS	6
ANOMALOUS BEHAVIOR OF FINITE ELEMENT FORMULATIONS	9
APPLICATIONS	15
REFERENCES	22

## TABLES

I. SOLUTIONS TO FREE OSCILLATOR PROBLEM (DISPLACEMENT/VELOCITY)	14
..	
II. SOLUTION TO $u + \dot{u} = H(t-1/2)$	16
..	
III. SOLUTION TO $u + \dot{u} = \delta(t-1/2)$	16

## LIST OF ILLUSTRATIONS

1. Divergent finite element solutions to free oscillator problem.	24
2. Displacement of beam at location of moving mass.	25

ACKNOWLEDGMENT

The author is especially grateful for the interest and assistance of Mr. Royce Soanes of the Benet Computer Science Laboratory.

## INTRODUCTION

According to Finlayson and Scriven<sup>1</sup> it is not variational notation or even the concept of a varied path which is the key criterion of a true variational 'principle', but rather the existence of a functional which when varied and set to zero, generates the governing equations and constraints for a given class of problems. In this sense, certain fundamental principles of mechanics such as d'Alembert's Principle do not truly qualify as variational principles. That is to say, these mechanical principles or 'laws' cannot be posed as central problems of the calculus of variations. On the other hand there are others, such as Hamilton's principle which do qualify as true variational principles. Yet it is d'Alembert's Principle which forms a basis for all analytical mechanics<sup>2</sup> and it follows, therefore, that the vanishing of the first variation of some functional is not a necessary condition for the scalar formulation of any mechanics problem - however elegant or convenient this may be.

Whether a true variational principle or a more fundamental variational statement is used to obtain a numerical solution to a dynamics problem, an important argument is that well established laws such as d'Alembert's Principle or true principles such as Hamilton's, are physically based and avoid the arbitrariness inherent in general weighted residual methods and contrived variational principles. Only variational principles which are also maximum or minimum principles appear to offer any advantage for obtaining

---

<sup>1</sup>Finlayson, B. A. and Scriven, L. E., "On the Search for Variational Principles," *Int. J. Heat Mass Transfer*, Vol. 10, 1967, p. 799-821.

<sup>2</sup>Lanczos, C., The Variational Principles of Mechanics, 3rd Edition, University of Toronto Press, 1966, pp. 70-72.

approximate solutions - mainly through their ability to provide bounds on the variational integral. Even then the system treated must be positive-definite and the upper and lower bounds are often too far apart to be of practical value. In brief, there seems to be little point in contriving a variational principle in preference to a variational law of mechanics despite the more primitive status of the latter. Indeed the many solutions to initial value dynamics problems achieved by C. Bailey<sup>3</sup> by applying the Ritz method to Hamilton's 'law of varying action' demonstrate the usefulness of variational formulations not qualifying as 'principles'. Thus motivated, the work herein explains the numerical difficulties encountered in attempting to generalize Bailey's formulations according to the method of finite elements.

#### FINITE ELEMENTS IN TIME

The many solutions achieved by C. Bailey were generated by the Ritz method<sup>4</sup> using a power series approximation in which globally defined polynomials are the basis functions. Ultimately the length of interval over which solutions may be generated as well as the detail to be provided in any subinterval will be limited by the degree of polynomial used as a basis. The pitfalls of using higher powered polynomials are well documented<sup>5</sup> and partially account for the use of locally (piecewise) defined basis functions (finite elements) to solve problems in many branches of mathematical physics. The

---

<sup>3</sup>Bailey, C. D., "The Method of Ritz Applied to the Equation of Hamilton," Computer Methods in Applied Mechanics and Engineering, 7, 1976, pp. 235-247.

<sup>4</sup>Kantorovich, L. V., and Krylov, V. I., Approximate Methods of Higher Analysis, Interscience Publishers, Inc., 1964, pp. 258-303.

<sup>5</sup>Conte, S. D., and de Boor, C., Elementary Numerical Analysis: An Algorithmic Approach, 2nd Edition, McGraw Hill, 1972, pp. 231-233.

extraordinary accuracy and simplicity of procedure attained by Bailey, however, is not to be understated.

Apart from avoiding the problems which can arise when higher powered polynomials are employed as basis functions, finite element formulations have other advantages when used to solve problems in continuum mechanics. Even though the principal motivation for their use has been the need to handle complicated boundary shapes (non-existent in the time domain) finite elements are also well suited to handle sudden changes in load functions, extending the interval of solution indefinitely without restart, and providing great detail to the solution in any subinterval. Thus despite the reservations expressed by Zienkowicz,<sup>6</sup> the extension of the finite element method to the solution of transient field problems is well motivated and was first reported by Argyris and Sharpf<sup>7</sup> and later by Fried.<sup>8</sup>

Both of these works attempt to use Hamilton's principle as a starting point for the finite element formulation of initial value problems. As will be pointed out in the following section, this cannot be accomplished without some logical inconsistency when bringing the initial data into the formulation. In the sequel it will be shown that the use of Hamilton's 'law', rather than Hamilton's 'principle', makes possible the logical incorporation of the initial conditions into the variational formulation.

---

<sup>6</sup>Zienkiewicz, O. C., The Finite Element Method, 3rd Edition, McGraw-Hill, 1977, pp. 569-70.

<sup>7</sup>Argyris, J. H., and Scharpf, D. W., "Finite Elements in Time and Space," Nuclear Engineering and Design, 10, 1969, 456-464.

<sup>8</sup>Fried, I., "Finite-Element Analysis of Time-Dependent Phenomena," AIAA Journal, 7, No. 6, pp. 1170-1172.

## HAMILTON'S PRINCIPLE - A CONSTRAINED VARIATIONAL PRINCIPLE

The following equation is known as the generalized principle of d'Alembert:<sup>9</sup>

$$\sum_{i=1}^N (\mathbf{F}_i - \dot{\mathbf{p}}_i) \cdot \delta \mathbf{r}_i = 0 \quad ; \quad (\dot{\phantom{r}}) = \partial / \partial t \quad (1)$$

This equation applies to any system of  $N$ -particles, the  $i$ th particle having a position  $\mathbf{r}_i$ , a momentum  $\dot{\mathbf{p}}_i$ , and subject to a resultant applied force  $\mathbf{F}_i$ .

Under the assumption that the virtual work of the applied forces is derivable from a scalar  $V$ , a time integration of equation (1) leads to Hamilton's law of varying action:<sup>10,11</sup>

$$\delta \int_{t_1}^{t_2} (T - V) dt - \sum_{i=1}^N m_i \dot{\mathbf{r}}_i \cdot \delta \dot{\mathbf{r}}_i \Big|_{t_1}^{t_2} = 0 \quad (2a)$$

$T$  is the kinetic energy of the system

$$T = 1/2 \sum_{i=1}^N m_i \dot{\mathbf{r}}_i \cdot \dot{\mathbf{r}}_i$$

and  $V$  is the potential energy of the forces impressed on the  $N$ -particles. The existence of  $V$  makes little difference as far as numerical calculations are concerned. In the event  $V$  does not exist, equation (2a) can be written:

$$\int_{t_1}^{t_2} (\delta T + \delta W) dt - \sum_{i=1}^N m_i \dot{\mathbf{r}}_i \cdot \delta \dot{\mathbf{r}}_i \Big|_{t_1}^{t_2} = 0 \quad (2b)$$

---

<sup>9</sup>Mierovitch, L., Methods of Analytical Dynamics, McGraw-Hill, 1970, p. 65.

<sup>10</sup>Bailey, C. D., "Application of Hamilton's Law of Varying Action," AIAA Journal, Vol. 13, No. 9, pp. 1154-1157.

<sup>11</sup>Hamilton, W. R., "Second Essay on a General Method in Dynamics," Philosophical Transactions of the Royal Society of London, 1835, pp. 95-144.

The bar signifies that in general the virtual work of the applied forces cannot be derived from any scalar function of the generalized coordinates. Either of equations (2) can be used as a basis for a Ritz approximation to a dynamics problem.

If  $\delta \bar{r}_i(t_1)$  and  $\delta \bar{r}_i(t_2)$  vanish in equation (2a), the result is Hamilton's principle:

$$\delta \int_{t_1}^{t_2} (T-V) dt = 0 \quad (3)$$

Since the vanishing of the displacement variations at the end points is not the only means by which the partial sum in equation (2a) may vanish, equation (3) may not always represent Hamilton's principle in the strict sense. Should equation (3) be used as a basis for the numerical solution of a dynamics problem without the requirement that all of the  $\delta \bar{r}_i$  vanish at  $t_1$  or  $t_2$ , zero momentum conditions will prevail instead as natural boundary conditions on those displacements whose variations are free. This aspect of variational principles is covered very clearly in many references (cf. ref. 12). An observation to be made here is that equation (3) corresponds to a system of boundary value problems - not initial value problems - since the partial sum can only vanish through boundary (end point) constraints either natural or imposed. Thus equation (3) cannot, with complete logic, be used to formulate any system of initial value problems of dynamics. The introduction of initial data has in fact always been the obstacle preventing the use of Hamilton's

---

<sup>12</sup>Courant, R., "Variational Methods for the Solution of Problems of Equilibrium and Vibrations," Bulletin of American Mathematical Society, 49, pp. 1-23.

principle for the variational formulation of initial value problems.<sup>13,14</sup>

Since equation (3) is a valid physical statement of mechanics only when the boundary constraints are such that the partial sum vanishes, it is proper to refer to this equation as a 'constrained variational principle' as opposed to equations (2) which are unconstrained variational laws of mechanics, suitable for the application of arbitrary constraint conditions.

#### GLOBAL AND PIECEWISE RITZ APPROXIMATIONS

Equations (2) and (3) differ only in the presence or absence of boundary terms. For the case of a single particle ( $N=1$ ) having only one degree of freedom  $u(t)$ , the Ritz procedure when applied to either of equations (2) leads to a scalar relation of the form:

$$\underset{\sim}{\delta} \underset{\sim}{U}^T \underset{\sim}{[} (K-B) \underset{\sim}{U} - F \underset{\sim}{] } = 0 \quad (4)$$

whereas for equation (3):

$$\underset{\sim}{\delta} \underset{\sim}{U}^T \underset{\sim}{[} K \underset{\sim}{U} - F \underset{\sim}{] } = 0 \quad (5)$$

Equations (4) and (5) are assumed to derive from applying the Ritz procedure whereby the displacement function  $u(t)$  is approximated as:

$$u(t) = \underset{\sim}{a}^T(t) \underset{\sim}{U} \quad (6)$$

The relation (6) applies to the entire interval of solution when globally defined basis functions are used or to a particular subinterval thereof when piecewise functions (finite elements) are employed. When a global power series approximation is used  $U$  is a vector of generalized coordinates, the first

---

<sup>13</sup>Tiersten, H. F., "Natural Boundary and Initial Conditions From a Modification of Hamilton's Principle," J. of Math. Physics, Vol. 9, No. 9, pp. 1445-1450.

<sup>14</sup>Gurtin, M. E., "Variational Principles for Linear Elastodynamics," Archive Ratl. Mech. Anal. 16, 34-50 (1964).

two of which are identifiable as  $u(t_1)$  and  $u(t_2)$ . The 'shape function',  $\tilde{a}(t)$ , in this case is simply:

$$\tilde{a}^T(t) = [1, t, t^2, \dots, t^n] , \quad t_1 \leq t \leq t_2 \quad (7)$$

If piecewise cubic Hermite polynomials are used instead, the components of  $\tilde{U}$  are local values of  $u$  and  $u'$  defined at the endpoints of a particular subinterval, and

$$\tilde{a}^T(t) = [2\tau^3 - 3\tau^2 + 1, \quad h(\tau^3 - 2\tau^2 + \tau), \quad 3\tau^2 - 2\tau^3, \quad h(\tau^3 - \tau^2)] \quad (8)$$

where  $\tau = t/h$ ,  $h$  being the length of the particular subinterval. Referring first to equation (5), it is noted that  $\tilde{K}$  tends to be singular of degeneracy one. For certain simple problems  $\tilde{K}$  may compute to be exactly singular. In general, however,  $\tilde{K}$  will only become singular in the limit as the number of basis functions employed in the Ritz approximation becomes infinite. The degeneracy of  $\tilde{K}$  represents the possibility that neither  $u(t_1)$  or  $u(t_2)$  has been specified. That is, if neither  $\delta u(t_1)$  or  $\delta u(t_2)$  vanishes, then  $\mu$  must vanish at both endpoints as natural boundary conditions. Under these conditions  $u(t)$  may only be determined to within an arbitrary constant. Thus in equation (5)  $\tilde{K}$  may only be reduced to a nonsingular matrix by specifying values for  $u(t_1)$  and/or  $u(t_2)$  so that the variations of one or both of these quantities vanish. The essence of the discussion which follows is not changed if, in the sequel, it is assumed that  $u(t_1)$  has been specified. This is known as a 'geometric' or 'imposed' constraint. Because  $\delta U_1 \equiv \delta u(t_1) = 0$  multiplies the first row of  $\tilde{K}$  in equation (5), this row is effectively removed from the formulation. Since the remaining variations are arbitrary the final set of equations to be solved is then:

$$\sum_{j=2}^n K_{ij}U_j = F_i - K_{i1}U_1, \quad i = 2, 3, \dots, n \quad (9)$$

where  $U_1 = u(t_1)$  is the specified value and  $n$  is the dimension of  $K$ . Whether these equations derive from a global power series approximation or from one based on finite elements, one may readily verify that as  $n$  is increased their solutions do indeed converge to the exact solution of the corresponding two point time-boundary value problem. Should one wish a solution to an initial value problem, however, equation (4) must be used instead of equation (5). In this case, specifying values for  $u(t_1)$  and  $\dot{u}(t_1)$  cause  $\delta U_1$  and  $\delta U_2$  to vanish thereby deleting the first two equations of this set. The resulting system of equations to be solved is thus:

$$\sum_{j=3}^n (K_{ij} - B_{ij})U_j = F_i - (K_{i1} - B_{i1})U_1 - (K_{i2} - B_{i2})U_2, \quad i = 3, 4, \dots, n \quad (10)$$

In all cases attempted to date, solutions to equations (10) have been observed to converge to the exact solution if these equations are derived using a global power series approximation but not if they are formulated by finite elements. An example of this anomaly will be given in the next section. As the only difference between equations (4) and (5) is a subtraction of  $B$  in the former, and in as much as convergence is achieved when equation (4) derives from a power series approximation, one suspects that it is the finite element representation of the matrix  $B$  which is somehow at fault. It is therefore of interest to know in more detail just how the subtraction of  $B$  is supposed to affect the coefficient matrix of the system.

In contrast to the matrix  $K$ , the matrix  $K-B$  must tend to be singular of degeneracy two - no constraints having been assumed a priori. Thus when  $u(t_1)$

is specified and the first row of  $\tilde{K}-\tilde{B}$  is deleted, the remaining equations still must possess one degeneracy in the limit as the number of basis functions becomes infinite. Thus the effect of subtracting  $\tilde{B}$  must be to free the natural boundary condition at  $\tilde{t}_2$  (inherent in equation (5)) and to introduce a degeneracy. This remaining degeneracy can only be removed by specifying the value of  $u(t)$  at a time other than  $\tilde{t}_1$  or a value for  $u$ , resulting in the deletion of another row of  $\tilde{K}-\tilde{B}$ .

#### ANOMALOUS BEHAVIOR OF FINITE ELEMENT FORMULATIONS

The degree to which the subtraction of the matrix  $\tilde{B}$  from  $\tilde{K}$  can both free the natural boundary condition at  $\tilde{t}_2$  and introduce a degeneracy differs with the type of approximation employed. When global power series approximations are used the  $\tilde{B}$  matrix is quite full and the subtraction affects many rows of  $\tilde{K}$ . When locally defined Hermite polynomials are used, however,  $\tilde{B}$  is very sparse and in fact contains only two non-zero components. Moreover, one of these appears in the first row of  $\tilde{B}$  which is deleted when  $u(\tilde{t}_1)$  is specified. In this case freeing the natural boundary condition and introducing a degeneracy depends on the subtraction from a single component of  $\tilde{K}$ . Even though both effects may actually be produced in the limit as the number of elements becomes infinite, the degree to which they are approximated for any finite number of elements is evidently insufficient and the solutions do not converge to the correct result. This is exemplified in Figure 1. The problem represented is that of a free oscillator of unit mass and stiffness, subject to the prescribed initial constraints of zero displacement and unit velocity. For this case, equation (2a) reads:

$$\int_0^\pi (\dot{u}\delta\dot{u} - u\delta u) dt - [\dot{u}\delta u]_0^\pi = 0 \quad (11)$$

or simply,

$$\int_0^{\pi} (u+u) \delta u dt = 0 \quad * \quad (12)$$

The finite element results of Figure 1 were obtained using piecewise cubic Hermite polynomials. (Higher ordered Hermite polynomials yield similar results.) It is observed that the solutions tend to diminish from the exact solution,  $\sin(t)$ , as the number of elements is increased. Using only two finite elements the finite element matrix formulation (equation (4)) for this problem is as follows:

$$0 = \delta U^T [K-B] U = [\delta U_1 \ \delta U_2 \ \delta U_3 \ \delta U_4 \ \delta U_5 \ \delta U_6] \cdot$$

$$\begin{array}{cccc|cc|ccccc|c}
 k_{11} & k_{12} & k_{13} & k_{14} & 0 & 0 & 0 & -1 & 0 & 0 & 0 & 0 & U_1 \\
 k_{21} & k_{22} & k_{23} & k_{24} & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & U_2 \\
 k_{31} & k_{32} & k_{33+k_{11}} & k_{34+k_{12}} & k_{13} & k_{14} & 0 & 0 & 0 & 0 & 0 & 0 & U_3 \\
 k_{41} & k_{42} & k_{43+k_{21}} & k_{44+k_{22}} & k_{23} & k_{24} & 0 & 0 & 0 & 0 & 0 & 0 & U_4 \\
 0 & 0 & k_{31} & k_{32} & k_{33} & k_{34} & 0 & 0 & 0 & 0 & 0 & 1 & U_5 \\
 0 & 0 & k_{41} & k_{42} & k_{43} & k_{44} & 0 & 0 & 0 & 0 & 0 & 0 & U_6
 \end{array} - \cdot \quad (13)$$

\*Note that Eq. (12) would also result from application of the Galerkin procedure, implying that the Galerkin method has physical justification for problems in dynamics.

Using expression (8), the element matrix  $\mathbf{k}$  is calculated in terms of the element length  $h$  as:

$$\mathbf{k} = \int_0^h (\mathbf{a}\mathbf{a}^T - \mathbf{a}\mathbf{a}^T) dt = \boxed{\begin{matrix} 
 \frac{6}{5h} - \frac{13h}{35} & \frac{1}{10} - \frac{11h^2}{210} & \frac{9h}{70} & \frac{6}{5h} & \frac{13h^2}{420} + \frac{1}{10} \\
 \frac{2h}{15} - \frac{h^3}{105} & - \frac{13h^2}{420} & - \frac{1}{10} & \frac{h^3}{140} & - \frac{h}{30} \\
 - \text{SYMM.} - & & & \frac{6}{5h} - \frac{13h}{35} & \frac{11h^2}{210} - \frac{1}{10} \\
 & & & \frac{2h}{15} - \frac{h^3}{105} & 
 \end{matrix}}$$

Since  $U_1$  is specified the first row of  $\mathbf{K} - \mathbf{B}$  is deleted. As the subtraction of  $\mathbf{B}$  only affects one row of the reduced system, the only way in which a degeneracy can be introduced is for the next to last row to join the space defined by the rows remaining. Thus rows two through six in equation (13) ideally would become linearly dependent. This dependency among rows must be quite general as specification of any other of the  $U_i$  must remove it.

One suspects that a simple subtraction of unity from  $K_{56}$  in equation (13) may not do the best job of introducing a degeneracy or of freeing the natural boundary condition at  $t_2 = \pi$ . One can gain some idea of how 'close' this subtraction brings the fifth row into the space of rows 2,3,4 and 6 by comparing it with its projection onto this space. Substituting  $\pi/2$  for  $h$ , the fifth row of equation (13) calculates to be:

---

\*All mathematics herein were performed using the MACSYMA (Project MAC's SYmbolic MAnipulation) system developed by the Mathlab Group of the MIT Laboratory for Computer Science.

[0.0 0.0 -0.96590326 -0.17637194 0.180505097 -0.970755175]

whereas its projection is:

[7.8587183E-3 -8.5978979E-3 -0.974496335

-0.184380835 0.172642875 -0.9617834].

Further calculations show that if the interval of solution remains fixed and the number of finite elements is allowed to increase, closer agreement between the next to last row vector and its projection is observed but this is not accompanied by a convergence of the solution vector toward the exact solution to the problem. While the exact reasons for this instability are not known it is apparent that the rate at which the next to last row tends to become dependent is important. It stands to reason, therefore, that should one invoke the limit condition without actually proceeding to the limit, a convergent sequence may result and indeed this proves to be the case.

Asserting that the row vectors two through six are linearly dependent allows the fifth row (equation) of equations (13) to be replaced by a linear combination of the others. For example, let

$$\tilde{R}_5 = \alpha_2 \tilde{R}_2 + \alpha_3 \tilde{R}_3 + \alpha_4 \tilde{R}_4 + \alpha_6 \tilde{R}_6 \quad (14)$$

where  $\tilde{R}_i$  denotes the  $i^{\text{th}}$  row of  $\tilde{K} - \tilde{B}$ . After imposing the second initial constraint,  $U_2 = 1$ , equations (13) can be written:

$$\delta \tilde{U}_3 \tilde{R}_3 \cdot \tilde{U} + \delta \tilde{U}_4 \tilde{R}_4 \cdot \tilde{U} + \delta \tilde{U}_5 (\alpha_2 \tilde{R}_2 + \alpha_3 \tilde{R}_3 + \alpha_4 \tilde{R}_4 + \alpha_6 \tilde{R}_6) \cdot \tilde{U} + \delta \tilde{U}_6 \tilde{R}_6 \cdot \tilde{U} = 0 \quad (15)$$

Since all variations in equation (15) are arbitrary, there results the following system of equations for solution:

$$0 = \tilde{R}_3 \cdot \tilde{U} = \tilde{R}_2 \cdot \tilde{U} = \tilde{R}_4 \cdot \tilde{U} = \tilde{R}_6 \cdot \tilde{U} \quad (16)$$

Thus the second equation (row) which was originally deleted through the specification of  $U_2$ , is brought back into the formulation in place of the fifth in a logical and consistent manner. Equations (16) are the same set as would result from following the procedure of Argyris and Scharpf. These authors, however, started with Hamilton's principle which requires that  $\delta U_1 = \delta U_5 = 0$ . This would delete the first and fifth equations from the set. Further specification of  $U_2$  should then delete the second equation as well, overspecifying the problem. Argyris and Scharpf<sup>7</sup> allow this equation to remain without justification. Moreover, no explanation is given as to why  $\delta U_5$  should vanish as  $U_5$  is never specified in an initial value problem. All of these inconsistencies derive from the fact that Hamilton's principle corresponds only to boundary value problems - never to initial value problems.

In summary, the work of this section shows that Hamilton's law of varying action, unlike Hamilton's principle, is an unconstrained variational statement permitting the introduction of arbitrary constraints including data ordinarily given for initial value problems. When piecewise Hermite cubic polynomials are used as a basis for a finite element formulation, the singular state of the resulting coefficient matrix in the limit justifies retention of the second equation of the system in preference to the next to last when typical initial values for displacement and velocity are specified. Following this procedure, convergent solutions are then obtained for the problem of the free oscillator considered in this section. These results are presented in Table I for formulations based on one, two, and six finite elements.

---

<sup>7</sup>Argyris, J. H., and Scharpf, D. W., "Finite Elements in Time and Space," Nuclear Engineering and Design, 10, 1969, 456-464.

TABLE I. SOLUTIONS TO FREE OSCILLATOR PROBLEM (DISPLACEMENT/VELOCITY)

0 < t <  $\pi$ 

6t/ $\pi$	One Element	Two Elements	Six Elements	Exact Solution
0	0.0*	0.0*	0.0*	0.0
	1.0*	1.0*	1.0*	1.0
1			0.49978005 0.86602547	0.5 0.86602541
2			0.86564452 0.50000025	0.86602541 0.5
3		0.97817298 2.02985945E-4	0.99956036 4.4572957E-7	1.0 0.0
4			0.86564496 -0.49999948	0.86602541 -0.5
5			0.499780823 -0.86602502	0.5 0.86602541
6	0.0166090783 -1.00079414	3.9845105E-4 -1.00000946	8.9120273E-7 -0.99999999	0.0 -1.0

\* Imposed values.

## APPLICATIONS

### Example 1. Linear Oscillator Subjected to Discontinuous Forces

A linear oscillator of unit mass and stiffness is subjected to a force  $f(t)$ . Two cases are considered:

$$(a) f(t) = H(t-1/2)$$

$$(b) f(t) = \delta(t-1/2)$$

$H$  and  $\delta$  are the Heaviside and Dirac functions respectively and for either of these cases equation (2) reads:

$$\int_{t_1}^{t_2} \{u\delta\dot{u} + (f(t)-u)\delta u\} dt - u\delta u \Big|_{t_1}^{t_2} = 0 \quad (17)$$

For case (a) four finite elements of equal length are used to approximate  $u(t)$  over the solution interval  $(0,2)$ . The element polynomial shape function is Hermite cubic and an element length of one half takes advantage of the specific shape of the forcing function. Table II compares the calculated displacements and velocities with those computed from the exact solution.

In case (b) a discontinuity in velocity can be expected in the solution. As the use of cubic shape functions enforces continuity of velocity throughout, a better solution might be expected when linear shape functions are employed. Table III compares the exact solution on the interval  $(0,1)$  with that obtained using ten such elements of equal length.

The two problems considered in this example demonstrate the manner in which the type of element and its points of attachment (i.e., the 'nodes' or 'grid points') may be varied to suit specified transient events.

TABLE II. SOLUTION TO  $u + u = H(t-1/2)$  $0 \leq t \leq 2.0$ 

t	Computed		Exact	
	Displacement	Velocity	Displacement	Velocity
0.0	0.0*	1.0*	0.0	1.0
0.5	0.47932149	0.87708716	0.47942555	0.877582565
1.0	0.96370936	1.0199163	0.96388844	1.01972786
1.5	1.45700388	0.91238744	1.45719267	0.91220819
2.0	1.83836447	0.5805616	1.83856024	0.58134814

\*Imposed values.

TABLE III. SOLUTION TO  $u + u = \delta(t-1/2)$  $0 \leq t \leq 1$ 

t	Computed Displacement	Exact Displacement
0.0	0.0*	0.0
0.1	0.1*	0.099833416
0.2	0.199001664	0.19866933
0.3	0.296016622	0.295520213
0.4	0.390076343	0.38941834
0.5	0.58007539	0.57925896
0.6	0.76428335	0.76331182
0.7	0.94086118	0.93973791
0.8	1.10804607	1.10677443
0.9	1.26416892	1.26275246
1.0	1.40767112	1.40611348

\*Imposed values.

### Example 2. Response of a Beam to a Moving Mass

A concentrated mass is assumed to move at constant velocity  $v$  along the length of a uniform Euler beam, simply supported at each of its ends and having zero displacement and velocity at  $t = 0$ . Under suitable definitions for  $k$  and  $m$ , the representative equations may be written:<sup>15</sup>

$$\begin{aligned} y'' + ky + f(x,t) &= 0 \\ y(0,t) = y''(0,t) = y(1,t) = y''(1,t) = y(x,0) = y(x,0) &= 0 \end{aligned} \quad (18)$$

The function  $f(x,t)$  consists of a sum of inertial terms:

$$f(x,t) = m(y + 2vy' + g + v^2y'')\delta(x-vt) \quad (19)$$

where  $g$  denotes the gravitational constant and  $\delta$  is the Dirac function. This problem is particularly interesting in that the conventional use of piecewise cubic shape functions to discretize the space variable only, introduces forces which are discontinuous functions of time into the resulting ordinary differential equations. These discontinuities are associated with the beam curvature load term appearing in the expression (19). Since the piecewise cubic polynomials are discontinuous in the second derivative at the element attachments, the term  $mv^2y''\delta(x-vt)$  - when multiplied by the shape function  $a(x)$  and integrated over the element length - will produce functions of time which are discontinuous whenever the moving mass arrives at any point of attachment. Clearly these discontinuities have nothing to do with the physics of the problem and are certain to invite trouble when one attempts to numerically inte-

---

<sup>15</sup>Simkins, T. E., "Unconstrained Variational Statements for Initial and Boundary-Value Problems," AIAA Journal, Vol. 16, No. 6, June 1978, pp. 559-563.

grate the time dependent equations via established algorithms. It is possible, of course, to use shape functions of higher degree to discretize the space variable thus eliminating the discontinuities at the onset but this is hardly consistent with the finite element method which should permit the use of even linear shape functions if need be. One is tempted to somehow 'smooth' these discontinuities, yet this should not be done in a purely arbitrary fashion. Integrating the effects of these forces throughout the time domain through the use of Hamilton's law of varying action provides a consistent way to handle this problem.

While it is possible to handle the space and time finite element discretizations in one operation, the amount of computation and computer programming tend to become inordinately large. Moreover, there exist any number of finite element codes (e.g. NASTRAN) which can quickly accomplish much of the space discretization. It seems more efficient, therefore, to apply the finite element method in two steps, by first discretizing the space variable and then applying Hamilton's law to the resulting system of ordinary differential equations in time. For the case at hand, the differential equations governing the motion of the  $i^{\text{th}}$  beam element turn out to be:

$$(\overset{\sim}{p} + \overset{\sim}{m} \overset{\sim}{c}_1) \overset{\sim}{u} + \overset{\sim}{m} \overset{\sim}{c}_2 \overset{\sim}{u} + (\overset{\sim}{q} + \overset{\sim}{m} \overset{\sim}{c}_3) \overset{\sim}{u} + \overset{\sim}{m} g a(\overset{\sim}{v} t) = 0 \quad (20)$$

$\overset{\sim}{p}$  and  $\overset{\sim}{q}$  are proportional to the usual mass and stiffness matrices for beam elements and have been evaluated many times in the literature. Here all of the beam elements are of the same length  $\ell$ , and the displacement within the  $i^{\text{th}}$  element is interpolated from  $\overset{\sim}{u}^i(t)$ , a vector of end point displacements and velocities, i.e.,

$$y(x,t) = \underset{\sim}{a}^T(\xi^i) \underset{\sim}{u}^i(t)$$

$$0 \leq \xi^i \leq 1 \quad (21)$$

where  $\xi^i(x) = x/\ell - (i-1)$ , a nondimensional element coordinate.

The  $\underset{\sim}{c}$  matrices in equations (20) correspond to transverse, Coriolis, and centrifugal accelerations respectively and are defined for the  $i^{\text{th}}$  element as follows:

$$\underset{\sim}{c}_1 = \underset{\sim}{a}(\xi^i) \underset{\sim}{a}^T(\xi^i) \Big|_{x=vt}$$

$$\underset{\sim}{c}_2 = 2va(\xi^i) \underset{\sim}{a}'^T(\xi^i) \Big|_{x=vt} \quad (22)$$

$$\underset{\sim}{c}_3 = v^2 \underset{\sim}{a}(\xi^i) \underset{\sim}{a}''^T(\xi^i) \Big|_{x=vt}$$

It is noted that  $\underset{\sim}{c}_3$  will be discontinuous at  $\xi^i = 0$  and  $\xi^i = 1$ . The function  $\underset{\sim}{m}$  takes on the value of  $m$  only when the concentrated mass lies within the  $i^{\text{th}}$  element, otherwise  $\underset{\sim}{m}$  is zero.

The element equations (20) are combined in the usual way to form  $N$  equations of motion for the combined structure. Symbolically:

$$\underset{\sim}{M}(t) \underset{\sim}{U} + \underset{\sim}{C}(t) \underset{\sim}{U} + \underset{\sim}{K}(t) \underset{\sim}{U} = \underset{\sim}{F}(t) \quad (23)$$

Each of the matrices in equation (23) can be viewed as a conventional matrix of constant coefficients plus a time variant set of components which are active in a band along its main diagonal as the moving mass traverses the beam in time. For this system of equations Hamilton's law of varying action can be written:

$$\sum_{i=1}^N \sum_{j=1}^N \left[ \int_{t_1}^{t_2} \{ \delta \dot{U}_i M_{ij} \dot{U}_j + \delta U_i [ (M_{ij} - C_{ij}) \dot{U}_j - K_{ij} U_j + F_i ] \} dt - \delta U_i M_{ij} U_j \Big|_{t_1}^{t_2} \right] = 0 \quad (24)$$

It is interesting to observe the accuracy of solution which can be obtained from equation (24) using only two finite elements in space and two in time. A formulation using two elements in space results in a system of  $N=4$  ordinary differential equations in time once the geometric support constraints have been applied. A two element formulation of these four equations for the time domain, followed by the application of all initial constraints in the manner summarized in Section 5, gives a final system of sixteen linear algebraic equations for solution. Figure 2 compares this solution with the experimental results of Ayre, Jacobsen, and Hsu<sup>16</sup> and a conventional finite element solution using three elements in the space domain followed by a time-integration of the equations (28) by Hamming's predictor-corrector algorithm.<sup>17</sup> The mass velocity in this case is  $v = v^*/2$ , --- $v^*$  being the lowest velocity to cause resonance when the load is a moving weight only and the magnitude assigned to the moving mass is 25% of the total mass of the beam. (Other parametric values are the same as those in reference 16.) The displacements have been normalized with respect to the maximum deflection produced if the weight was applied statically at midspan and  $L$  is the total beam length. In particular one notes that the conventional solution obtained via three finite elements in space only, produces non-physical discontinuities in the slope of the solution curve at  $vt/L = 1/3, 2/3$ . (The continuous data

---

<sup>16</sup>Ayre, R. S., Jacobsen, L. S., and Hsu, C. S., "Transverse Vibration of One and of Two Space Beams Under the Action of a Moving Mass Load," Proceedings of First National Congress on Applied Mechanics, June 1951.

<sup>17</sup>Ralston and Wilf, Mathematical Methods for Digital Computers, Wiley and Sons, NY, London, 1960, pp. 95-109.

for generating this curve is obtained by interpolating the solution to equation (23) using equation (21).) No discontinuities of this sort can arise when finite elements in space and time are employed. Improved agreement with the experimental results is also observed.

## REFERENCES

1. Finlayson, B. A. and Scriven, L. E., "On the Search for Variational Principles," *Int. J. Heat Mass Transfer*, Vol. 10, 1967, pp. 799-821.
2. Lanczos, C., The Variational Principles of Mechanics, 3rd Edition, University of Toronto Press, 1966, pp. 70-72.
3. Bailey, C. D., "The Method of Ritz Applied to the Equation of Hamilton," *Computer Methods in Applied Mechanics and Engineering*, 7, 1976, pp. 235-247.
4. Kantorovich, L. V., and Krylov, V. I., Approximate Methods of Higher Analysis, Interscience Publishers, Inc., 1964, pp. 258-303.
5. Conte, S. D., and de Boor, C., Elementary Numerical Analysis: An Algorithmic Approach, 2nd Edition, McGraw Hill, 1972, pp. 231-233.
6. Zienkiewicz, O. C., The Finite Element Method, 3rd Edition, McGraw-Hill, 1977, pp. 569-70.
7. Argyris, J. H., and Scharpf, D. W., "Finite Elements in Time and Space," *Nuclear Engineering and Design*, 10, 1969, 456-464.
8. Fried, I., "Finite-Element Analysis of Time-Dependent Phenomena," *AIAA Journal*, 7, No. 6, pp. 1170-1172.
9. Mierovitch, L., Methods of Analytical Dynamics, McGraw-Hill, 1970, p. 65.
10. Bailey, C. D., "Application of Hamilton's Law of Varying Action," *AIAA Journal*, Vol. 13, No. 9, pp. 1154-1157.
11. Hamilton, W. R., "Second Essay on a General Method in Dynamics," *Philosophical Transactions of the Royal Society of London*, 1835, pp. 95-144.

12. Courant, R., "Variational Methods for the Solution of Problems of Equilibrium and Vibrations," Bulletin of American Mathematical Society, 49, pp. 1-23.
13. Tiersten, H. F., "Natural Boundary and Initial Conditions From a Modification of Hamilton's Principle," J. of Math. Physics, Vol. 9, No. 9, pp. 1445-1450.
14. Gurtin, M. E., "Variational Principles for Linear Elastodynamics," Archive Ratl. Mech. Anal. 16, 34-50 (1964).
15. Simkins, T. E., "Unconstrained Variational Statements for Initial and Boundary-Value Problems," AIAA Journal, Vol. 16, No. 6, June 1978, pp. 559-563.
16. Ayre, R. S., Jacobsen, L. S., and Hsu, C. S., "Transverse Vibration of One and of Two Space Beams Under the Action of a Moving Mass Load," Proceedings of First National Congress on Applied Mechanics, June 1951.
17. Ralston and Wilf, Mathematical Methods For Digital Computers, Wiley and Sons, NY, London, 1960, pp. 95-109.

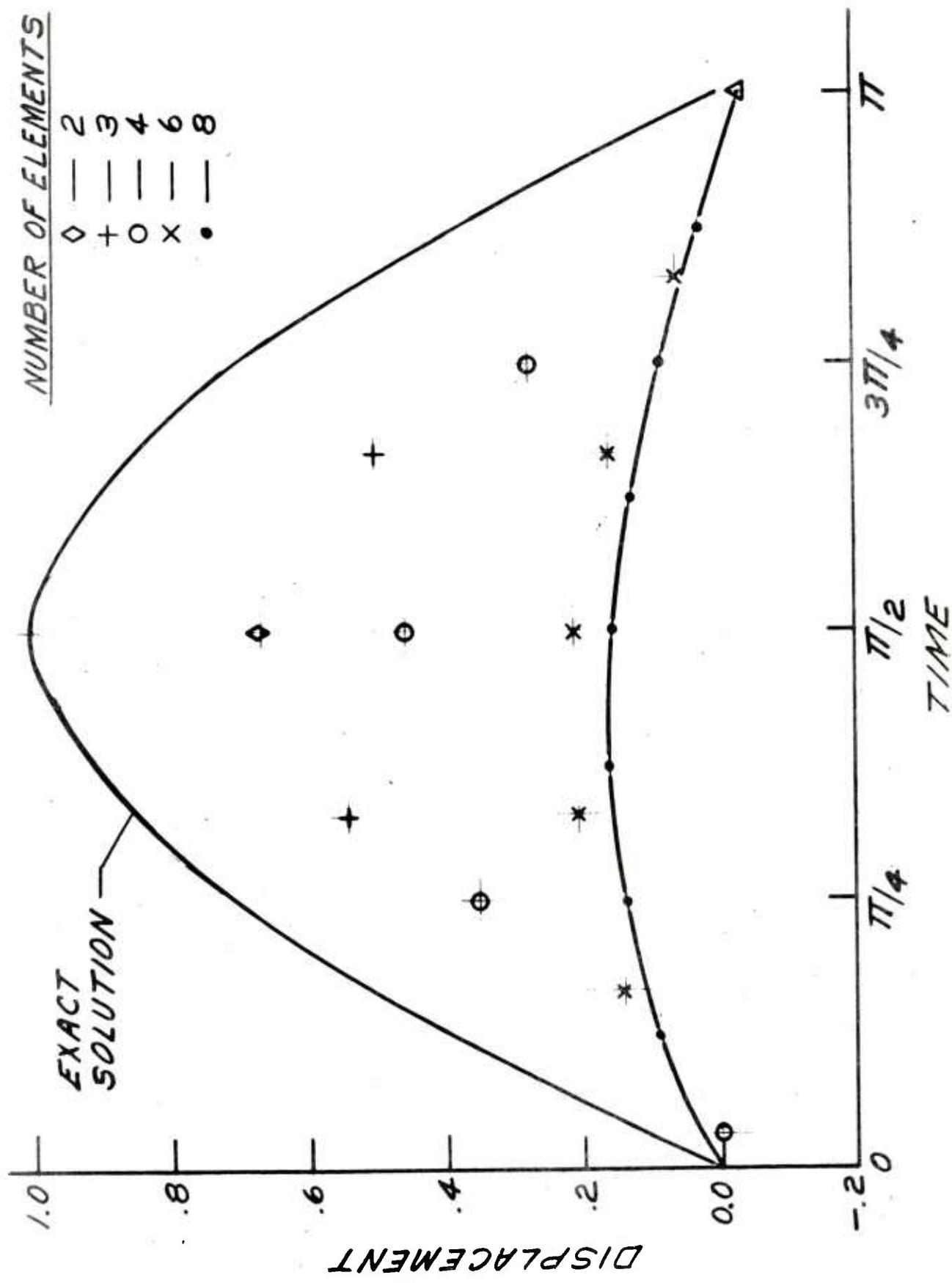


Figure 1. Divergent Finite Element Solutions to Free Oscillator Problems

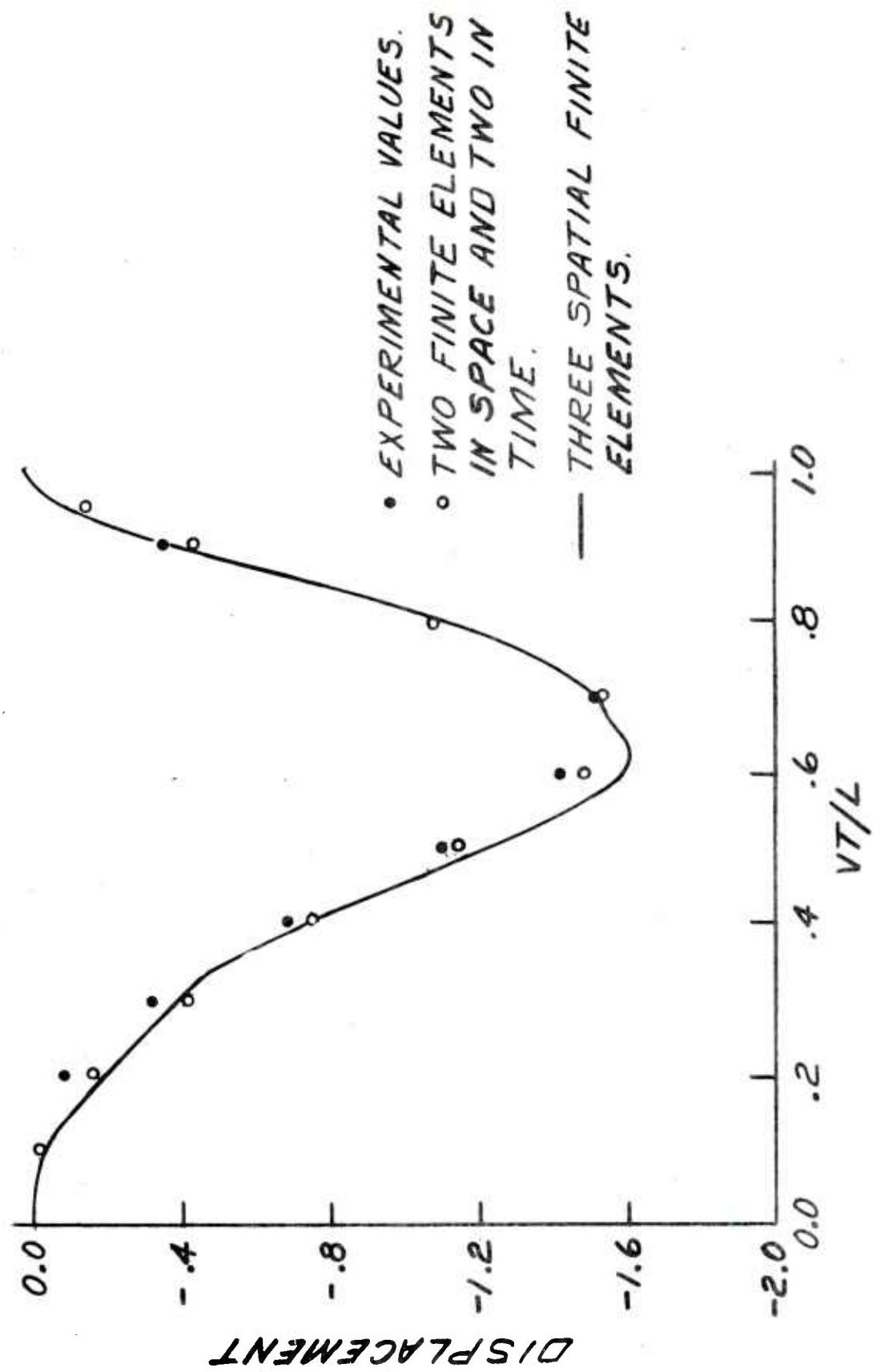


Figure 2, Displacement of Beam at Location of Moving Mass.

## TECHNICAL REPORT INTERNAL DISTRIBUTION LIST

	<u>NO. OF COPIES</u>
COMMANDER	1
CHIEF, DEVELOPMENT ENGINEERING BRANCH	1
ATTN: DRDAR-LCB-DA	1
-DM	1
-DP	1
-DR	1
-DS	1
-DC	1
CHIEF, ENGINEERING SUPPORT BRANCH	1
ATTN: DRDAR-LCB-SE	1
-SA	1
CHIEF, RESEARCH BRANCH	2
ATTN: DRDAR-LCB-RA	1
-RC	1
-RM	1
-RP	1
CHIEF, LWC MORTAR SYS. OFC.	1
ATTN: DRDAR-LCB-M	
CHIEF, IMP. 81MM MORTAR OFC.	1
ATTN: DRDAR-LCB-I	
TECHNICAL LIBRARY	5
ATTN: DRDAR-LCB-TL	
TECHNICAL PUBLICATIONS & EDITING UNIT	2
ATTN: DRDAR-LCB-TL	
DIRECTOR, OPERATIONS DIRECTORATE	1
DIRECTOR, PROCUREMENT DIRECTORATE	1
DIRECTOR, PRODUCT ASSURANCE DIRECTORATE	1

NOTE: PLEASE NOTIFY ASSOC. DIRECTOR, BENET WEAPONS LABORATORY, ATTN: DRDAR-LCB-TL, OF ANY REQUIRED CHANGES.

TECHNICAL REPORT EXTERNAL DISTRIBUTION LIST

	<u>NO. OF COPIES</u>		<u>NO. OF COPIES</u>
ASST SEC OF THE ARMY RESEARCH & DEVELOPMENT ATTN: DEP FOR SCI & TECH THE PENTAGON WASHINGTON, D.C. 20315		COMMANDER US ARMY TANK-AUTMV R&D COMD ATTN: TECH LIB - DRDTA-UL MAT LAB - DRDTA-RK WARREN MICHIGAN 48090	1
COMMANDER US ARMY MAT DEV & READ. COMD ATTN: DRCDE 5001 EISENHOWER AVE ALEXANDRIA, VA 22333	1	COMMANDER US MILITARY ACADEMY ATTN: CHMN, MECH ENGR DEPT WEST POINT, NY 10996	1
COMMANDER US ARMY ARRADCOM ATTN: DRDAR-LC -ICA (PLASTICS TECH EVAL CEN) -LCE -LCM -LCS -LCW -TSS(STINFO) DOVER, NJ 07801	1 1 1 1 1 1 1 2	COMMANDER REDSTONE ARSENAL ATTN: DRSMI-RB -RRS -RSM ALABAMA 35809 COMMANDER ROCK ISLAND ARSENAL ATTN: SARRI-ENM (MAT SCI DIV) ROCK ISLAND, IL 61202	2 1 1 1 1 1 1 1
COMMANDER US ARMY ARRCOM ATTN: DRSAR-LEP-L ROCK ISLAND ARSENAL ROCK ISLAND, IL 61299	1	COMMANDER HQ, US ARMY AVN SCH ATTN: OFC OF THE LIBRARIAN FT RUCKER, ALABAMA 36362	1
DIRECTOR US Army Ballistic Research Laboratory ATTN: DRDAR-TSB-S (STINFO) ABERDEEN PROVING GROUND, MD 21005.	1	COMMANDER US ARMY FGN SCIENCE & TECH CEN ATTN: DRXST-SD 220 7TH STREET, N.E. CHARLOTTESVILLE, VA 22901	1
COMMANDER US ARMY ELECTRONICS COMD ATTN: TECH LIB FT MONMOUTH, NJ 07703	1	COMMANDER US ARMY MATERIALS & MECHANICS RESEARCH CENTER ATTN: TECH LIB - DRXMR-PL WATERTOWN, MASS 02172	2
COMMANDER US ARMY MCBILITY EQUIP R&D COMD ATTN: TECH LIB FT BELVOIR, VA 22060	1		

NOTE: PLEASE NOTIFY COMMANDER, ARRADCOM, ATTN: BENET WEAPONS LABORATORY, DRDAR-LCB-TL, WATERVLIET ARSENAL, WATERVLIET, N.Y. 12189, OF ANY REQUIRED CHANGES.

TECHNICAL REPORT EXTERNAL DISTRIBUTION LIST (CONT)

	<u>NO. OF COPIES</u>		<u>NO. OF COPIES</u>
COMMANDER US ARMY RESEARCH OFFICE. P. O. BOX 12211 RESEARCH TRIANGLE PARK, NC 27709		COMMANDER DEFENSE TECHNICAL INFO CENTER ATTN: DTIA-TCA CAMERON STATION ALEXANDRIA, VA 22314	12
COMMANDER US ARMY HARVEY DIAMOND LAB ATTN: TECH LIB 2800 POWDER MILL ROAD ADELPHIA, MD 20783	1	METALS & CERAMICS INFO CEN BATTELLE COLUMBUS LAB 505 KING AVE COLUMBUS, OHIO 43201	1
DIRECTOR US ARMY INDUSTRIAL BASE ENG ACT ATTN: DRXPE-MT ROCK ISLAND, IL 61201	1	MECHANICAL PROPERTIES DATA CTR BATTELLE COLUMBUS LAB 505 KING AVE COLUMBUS, OHIO 43201	1
CHIEF, MATERIALS BRANCH US ARMY R&S GROUP, EUR BOX 65, FPO N.Y. 09510	1	MATERIEL SYSTEMS ANALYSIS ACTV ATTN: DRXSY-MP ABERDEEN PROVING GROUND MARYLAND 21005	1
COMMANDER NAVAL SURFACE WEAPONS CEN ATTN: CHIEF, MAT SCIENCE DIV DAHLGREN, VA 22448	1		
DIRECTOR US NAVAL RESEARCH LAB ATTN: DIR, MECH DIV CODE 26-27 (DOC LIB) WASHINGTON, D. C. 20375	1		
NASA SCIENTIFIC & TECH INFO FAC. P. O. BOX S757, ATTN: ACQ BR BALTIMORE/WASHINGTON INTL AIRPORT MARYLAND 21240	1		

NOTE: PLEASE NOTIFY COMMANDER, ARRADCOM, ATTN: BENET WEAPONS LABORATORY, DRDAF-LCB-TL, WATERVLIET ARSENAL, WATERVLIET, N.Y. 12189, OF ANY REQUIRED CHANGES.